SIMPLIFIED METHOD FOR DETERMINING COMPRESSOR REQUIREMENTS FROM FIELD DATA

David J. Sager

In order to select a compressor, or calculate the performance of a compressor, certain design criteria must be known. This criteria is listed below:

- 1. <u>Gas analysis</u>. A gas analysis is preferred, but if one is not available, then the specific gravity or molecular weight, n value, critical pressure and critical temperature of the gas can be used.
- <u>Capacity</u>. This is generally given in either MSCFD or MMSCFD (Thousand Standard Cubic Feet Per Day or Million Standard Cubic Feet Per Day) at a given base pressure and temperature. (Example: 14.65 PSIA & 60°F).
- 3. Inlet temperature. At the compressor inlet.
- 4. Inlet pressure. At the compressor inlet.
- 5. Discharge pressure. At the compressor outlet.
- 6. Elevation. At the compressor location.

Assuming values for the above design criteria, calculations can be made to either select a compressor, or calculate the performance of an existing compressor. For example, assume the following design criteria:

1. Gas analysis: Mol %

Methane	84.9%
Ethane	8.2%
Propane	2.5%
Iso-Butane	0.5%
Norm-Butane	1.2%
Peutane	0.5%
Hexanes +	0.5%
Nitrogen	1.2%
Carbon Dioxide	0.5%
-	100.0%

- 2. Capacity = 2.0 MMSCFD (measured at 14.65 PSIA & 60°F).
- 3. Inlet temperature = $80^{\circ}F$
- 4. Inlet pressure = 240 PSIG
- 5. Discharge pressure = 1050 PSIG
- 6. Elevation = 2700 ft.

Based on the above C.O.S. (Conditions Of Service) we can determine the following values which are necessary to select a compressor, or calculate its performance.

- A. <u>Barometric pressure</u>. The barometric pressure is determined by the elevation. This can be arrived at by using a formula, or an existing curve of which there are many. (See Exhibit A.) By using the attached curve (Exhibit A), it can be determined that the barometric pressure at 2700 ft. is 13.3 PSIA. It is necessary to know the barometric pressure because all compressor calculations are made using absolute values, whether it be pressure or temperature. Absolute pressures are gauge pressures plus the barometric pressure. In this case, the absolute inlet pressure (Ps) is; Ps = 240 PSIG +13.3 PSI = 253.3 PSIA; and the absolute discharge pressure (Pd) is Pd = 1050 PSIG +13.3 PSI = 1063.3 PSIA.
- B. The ratio of compression. To determine the ratio of compression, the formula:

 $Rc = \frac{Pd (PSIA)}{Ps (PSIA)}$ is used, where Rc is the ratio of compression, Pd is the

absolute discharge pressure and Ps is the absolute inlet pressure. For our case the ratio of compression is:

 $Rc = \frac{1063.3 (PSIA)}{253.3 (PSIA)} = 4.20$

C. <u>The number of stages of compression required</u>. The number of stages of compression required is normally determined by the discharge temperature of the compressor. This can be either the manufacturers' maximum allowable discharge temperature limit, or a specified limit, such as 300^{0} F as in API - 11P.

Another determining factor of the number of stages required is the rod load or frame load rating of the compressor. In some cases you may not be exceeding the temperature limit, but you may be exceeding the allowable rod load limit of the compressor. By adding one more stage of compression you can reduce the rod load on the compressor. The calculations will follow.

Another determining factor for the number of stages is the volumetric efficiency of the cylinder. This comes into play mainly on propane applications because the low n value of propane allows a high ratio of compression without exceeding the discharge temperature limit, but this high ratio of compression causes a low volumetric efficiency and most manufacturers limit the V.E. to a minimum of 15%.

With all this, a good rule of thumb for the maximum number of ratios in a stage of compression is 4.0 to 4.5. Again, as long as the discharge temperature limit, the rod load limit, and the volumetric efficiency are not exceeded, then the ratio of compression in that stage is acceptable.

D. <u>Discharge temperature</u>. To determine the theoretical discharge temperature for a given ratio of compression, the following formula is used:

$$Td = Ts (Rc^{\frac{n-1}{n}})$$

Where Td is the absolute discharge temperature, Ts is the absolute inlet temperature and n is the ratio of specific heats of the gas. The absolute temperatures are in degrees Rankin, which is degrees Fahrenheit plus 460. n, Or the ratio of specific heats is determined by the gas analysis or from the specific gravity. (See Exhibit B where the n value 1.267 was calculated basis the gas analysis, and then compared that to Exhibit C using a temperature of 60° F which is what Exhibit B is based on.)

Using the value 1.267 for n, the discharge temperature for the Rc of 4.20, when Ts = 540 (80 + 460), is: $\frac{.267}{1.267}$ Td = 540 (4.20) = 731 - 460 = 271° F

An easier way to determine the discharge temperature is to use a nomograph. (See Exhibit D.) To use the nomograph, you first find the ratio of compression on the right hand side. Go left to the n (k) value, then go vertically (up or down) to the inlet temperature. At the point

Seeing that the Td is $271^{\circ}F$, we can do this compression in a single stage of compression, unless we find out later that we exceed the rod load or minimum V.E. limits of the compressor.

where you intersect the inlet temperature curve, read left to the discharge

E. Estimate the compressor horsepower required. In order to estimate the compressor horsepower required to compress 2.0 MMSCFD from Ps = 253.3 PSIA to Pd = 1063.3 PSIA, the following formula can be used.

Compressor bhp = (22) (Rc per stage) (No. of stages) (MMSCFD) or,

Compressor bhp = (22)(4.20)(1)(2.0) = 184.8 bhp

This equation is close for gases with a specific gravity around 0.65, and stage compression ratios of 2.5.

A quick and easy method to determine the required compressor horsepower is to use the chart on Exhibit E. To use this chart, find the suction pressure in the left hand column and read across to the correct vertical discharge pressure column. This will give you the horsepower for each 1.0 MMSCFD of gas being compressed. Multiply this horsepower figure by the capacity (in MMSCFD) to determine the total horsepower required. Example, Ps = 250 PSIG and Pd = 1000 PSIG which gives 85 bhp/MMSCFD. Multiplying 85 bhp/MMSCFD times 2.0 MMSCFD gives a horsepower of 170.0 bhp. This compares with the 184.8 bhp calculated by the above formula.

F. Determine the required compressor displacement. To do this, we must convert the capacity from MMSCFD to acfm (actual cubic feet per minute at suction conditions). For this example it is at 250 PSIG and 80°F.

The formula used for this conversion is:

acfm = $\frac{\text{SCFD}}{1440} \times \frac{\text{Base Pressure (PSIA)}}{\text{Inlet Pressure (PSIA)}} \times \frac{\text{Inlet Temperature (}^{O}\text{R}\text{)}}{\text{Base Temperature (}^{O}\text{R}\text{)}}$

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temperature.

$$acfm = \frac{2,000,000}{1440} \times \frac{14.65}{253.3} \times \frac{540}{520} = 83.4 cfm$$

Once the acfm is known, the required displacement can be estimated by using the formula:

Displacement (cfm) =
$$\frac{acfm}{V.E.}$$
 (Volumetric Efficiency)
or
Displacement (cfm) = $\frac{83.4}{0.61}$ = 136.7 cfm

The V.E. of 0.61 was obtained by using Exhibit F. Exhibit F lets you pick the approximate V.E. basis the known ratio of compression and an estimated cylinder clearance. For estimating purposes, we use a cylinder clearance of 15%. By reading up vertically on the 15% clearance line to a ratio of compression of 4.20, and then left to the volumetric efficiency scale, we get a volumetric efficiency of 61.0%.

At this point it is advisable to stop and summarize what we have determined regarding the selection and performance of the compressor for given C.O.S. of:

- 1. Gas n = 1.267
- 2. Capacity 2.0 MMSCFD
- 3. Inlet temperature 80° F
- 4. Inlet pressure 250 PSIG
- 5. Discharge pressure 1050 PSIG
- 6. Elevation 2700 ft.

The following has been determined about the compressor selection and performance:

- A. Barometric pressure 13.3 PSIA
- B. Rc 4.20
- C. Number of stages of compression one
- D. Theoretical discharge temperature 271°F
- E. Compressor brake horsepower 184.8
- F. Required cylinder displacement 136.7 cfm
- G. Volumetric efficiency 61.0%
- H. Required cylinder working pressure rating 1050 PSIG minimum, as this is the required discharge pressure.

At this point it is necessary to mention that without some specific technical data from a compressor manufacturer, it is difficult to finish sizing this compressor. The minimum technical data required to complete this sizing is the horsepower and frame load rating of available frames, and the cylinder bores for these different frames with their corresponding working pressure ratings and inherent clearances.

Since Compressor Systems, Inc., who I am associated with, is a franchised distributor and packager of several different compressor lines, much technical data is available to us to accurately size and calculate performance on these different compressors. Also, being familiar with these different lines of compressors, it is possible for us to rule out several lines due to the horsepower and the high discharge pressure (1050 PSIG). The decision then on the compressor selection would be for either a balanced opposed compressor with one or two cylinders (dictated by displacement and working pressure rating), or a single-throw horizontal compressor. Generally, we would look at a single cylinder compressor first to see if it could handle the given conditions of services.

In this example, the single-throw horizontal compressor will be looked at.

Knowing that the Worthington HBGG compressor line has the following frames available:

Stroke (Ins.)	<u> Piston Rod Dia. (Ins.)</u>	Max. B.H.P.	Max. Frame Load (Lbs.)
5	1.375	39	5,500
7	1.625	66	9,000
9	1.750	100	14,000
11	2.000	170	18,000
13	2,500	220	25,000

it can be determined from the estimated bhp of 184.8 that the 13" stroke frame is required for this application.

From Exhibit G (dimension and specification page form Worthington's brochure entitled "Type HBGG Single-Horizontal Gas-Field Compressors") it can be seen that the closest bore size on the 13" frame to handle this application is the 6" bore which has a working pressure rating of 1250 PSIG, but a displacement of only 127.0 cfm. This size may work even though it is estimated that 136.7 cfm is required, and that is due to the actual inherent miminum clearance on this cylinder being only 9.0% and not 15% as used in the approximation done earlier.

Working a little differently now to see if the 6 X 13 HBGG size will work on this sample compressor application, the following steps will be taken:

- Find V.E. basis actual cylinder clearance of 9.0%. From Exhibit F, reading up from 9.0% clearance to 4.20 ratios, and then to the left, the approximate V.E. is 74.-%.
- 2. <u>Calculate compressor acfm knowing the cylinder displacement is 127.0</u> cfm and the V.E. is 74.0%.

Displacement (cfm) = $\frac{acfm}{V.E.}$, then

acfm = Displacement (cfm) (V.E.) = (127) (.74) = 94.0

 Calculate MMSCFD to see if this compressor will handle the desired flow of 2.0 MMSCFD. Basis the formula: acfm = $\frac{SCFD}{1440} \times \frac{Base Pressure (PSIA)}{Suction Pressure (PSIA)} \times \frac{Suction Temperature (^{O}R)}{Base Temperature (^{O}R)}$

then

$$94.0 = \frac{\text{SCFD}}{1440} \times \frac{14.65}{253.3} \times \frac{540}{520}$$

SCFD = $\frac{(94.0) (1440) (253.3) (520)}{(14.65) (540)} = 2,253,707$

4. Calculate compressor bhp basis actual flow of 2,253,707 SCFD. Using the formula:

Compressor bhp = (22) (Ratios/Stage) (No. of stages) (MMSCFD)

Compressor bhp = (22) (4.20) (1) (2.253) = 208.2

5. Calculate frame load to see that it is less than the maximum allowable.

Two additional pieces of information are required to make the frame load calculation. The two items are the area of the niston, and the cross-sectional area of the piston rod. From Exhibit H (Areas of Circles) it shows that the area of a 6" circle is 28.3 square inches, and the area of a 2.5" circle is 4.9 square inches. Having these two areas it is possible to use the formula:

- F.L. compression = Area of Piston X Discharge Pressure (PSIA) -(Area of Piston-Area of Piston Rod) X Suction Pressure (PSIA)
- F.L. compression = (28.3) (1063.3) (28.3 4.9) (253.3) = 24,164 lbs.
- F.L. tension = (Area of Piston-Area of Piston Rod) X Discharge Pressure (PSIA) -Area of Piston X Suction Pressure (PSIA)

F.L. tension = (28.3 - 4.9)(1063.3) - (28.3)(253.3) = 17.713 lbs.

Both of these numbers should be positive to indicate that rod Note: reversal is taking place. Having a rod reversal indicates that proper lubrication is ocurring between the crosshead and crosshead pin.

Seeing that the frame load in both compression and tension is below 25,000 lbs. indicates that this compressor is acceptable for this particular application.

A check of this simplified sizing method is included as Exhibit I. The same conditions as were used in the sample problem were run on the Compressor Systems, Inc. computer, and the estimated performance from the computer is very encouraging for using this simplified method. The data fed into the computer was as follows:

1. Elevation.

- 2. Gas analysis.
- 3. Make of compressor.
- 4. Model of compressor.
- Number of stages of compression. 5.

- 6. Number of cylinders.
- 7. Cylinder diameter.
- 8. Cylinder clearance.
- 9. Suction temperature $({}^{O}F)$.
- 10. Suction pressure (PSIG).
- 11. Discharge pressure (PSIG).

A comparison of the computer calculated performance with the simplified version discussed above is shown below:

- 1. Barometric pressure (PSIA) was 13.296 from the computer vs. 13.3 (PSIA) from Exhibit A.
- 2. Discharge temperature (^OF) was 255.5 from the computer vs. 271 from the nomograph.
- 3. The compression ratio calculated by the computer was 4.1978 vs. 4.20 by hand.
- 4. The volumetric efficiency was 71.25% from the computer vs. 74.0% from Exhibit E.
- 5. Flow (capacity) was 2268 MSCFD or 2,268,000 SCFD vs. 2,253,707 which was calculated by the simplified method. (Note, C.S.I.'s computer program is based on using compressibility which is a relationship involving the critical temperature and pressure of the gas stream and this accounts for some of the difference between the computer calculation and the simplified version. For record purposes, the critical temperature used was 381^oR and the critical pressure was 670 PSIA.)
- 6. Total brake horsepower from the computer was 196.0 as compared to the 208.2 which was estimated from the simplified formula.
- 7. The computed compression and tension rod loads were 24,145 lbs. and 17,682 lbs., respectively. This compared to the estimated 24,164 lbs. and 17,713 lbs., respectively.

As can be seen by the above comparison, the simplified sizing/calculating method presented here is very reliable for predicting horsepower and displacement for individual well compressors. If enough data is known about a particular compressor, a good prediction of estimated performance can be made for any given set of conditions.



GAS	FORMULA		NOL.WT.	HOL. # FRACT.	CP/HOL.	Cø/NOL. FRACT.	TĊ	PART TC	PC	Pâr P.C
ethane	CH.	.849	16.042	13.62	8.435	7.16	344	292	673	571
thene 2%	С.н.	.082	30.068	2.47	12.304	1.01	550	45	709	58
ropane	с.н.	.025	44.094	1.10	17.086	0.43	666	17	618	15
e-Sutane	C.H	.005	58.120	0.29	22.463	0.11	733	4	543	3
orm. Butane	C.H	.012	58.120	0.70	23.079	0.28	767	9	529	6
ntane	C.H.,	.005	72.146	0.36	29.671	0.14	847	4	4 85	2
224fe	C.N	.005	86.172	0.43	34.331	0.17	914	. 5	433	2
eptane	C,H		100.198	•	39.999		973		397	
hene	С, Н.		28.052		10.127	[•	510		749	
opene	C.N.		42.078	l	15.190		657		668	
itene	с.н.		56.104		20.546		751		819	
intene	C.#++		70.130	1	26.649		854		595	
tzeve	C.H.,		84.156		30.296		93 I			
iter	N-0		18.02	-	8.03		1165		3206	
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itrogen	N,	.012	28.016	0.34	6.954	0.08	228	3	492	6
erbon Honoxide	co		28.010	•	6.958	1	242		514	}
arbon Dioxide	co,	.005	44.010	0.22	8.762	0.04	548	3	1073	5
yd. Sulfice	H · S		34.076		8.655		672	1	1306	
ulph. Dioxice	SO :	1	. 54.060	1	9,417	ł.	775	1	1142	
monia		i	E 17.030	,	8.856	i I	731		1657	
rgon		i 1 1	39.944		4.913		272	1	705	
ital Cs/Mol. 9.4	20	1.000	Hol.Wt.	19.53	Cp/Mol.= Minus Cv/Mol.=	9.420 <u>1.986</u> 7.434	Ťc	382	Pc	668

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Methane	сн ₄	.849	16.042	13.62	8.435	7.16	344	292
Ethane	с ₂ н ₆	.082	30.068	2.47	12.304	1.01	550	45
Propane	с ₃ н ₈	.025	44.094	1.10	17.086	0.43	666	17
Iso-Butane	C4H10	.012	58.120	0.29	22.463	0.11	733	4
Norm. Butane	C4 ^H 10	.005	58.120	0.70	23.079	0.28	767	9
Pentane	с ₅ н ₁₂	.005	72.146	0.36	28.671	0.14	847	4
Hexane	с ₆ н ₁₄		86.172	0.43	34.331	0.17	914	5
Heptane	с ₇ н ₁₆		100.198		39.999		973	
lithere			28 052		10 127		510	
ALINENE	^C 2 ⁿ 4		20.052		10.127		510	ļ
Propene	C_H.		42.078	}	15.190		657	

MOL. WT.

2.5%	Propane,	0.5%	Iso-Butane,	1.2% Norm.	Butane	,		
0.5%	Pentane,	0.5%	ilexanes +,	1.2% Nitrog	en, and	0.5%	Carbon	Dioxide

% MOL.

FORMULA

GAS

17 618 15 4 543 3 9 529 6 485 2 4 433 2 5 397 749 668 15.190 657 42.078 Propene ^с3^н6 C₄H₈ 56.104 20.646 751 619 Butene 70.130 26.649 854 595 Pentene $C_{4}^{H}10$ 931 84.156 30.296 Hexene C6H12 3206 H_0 18.02 8.03 1165 Water 74 306 Hydrogen 2.016 6.871 Н2 279 7.002 730 32.000 0xygen 02 .012 0.34 6.954 0.08 228 3 492 6 28.016 Nitrogen ^N2 Carbon Monoxide СО .005 28.010 6.958 242 514 1073 5 Carbon Dioxide 44.010 0.22 8.762 0.04 548 3 C0₂ 1306 672 34.076 8.655 Hyd. Sulfide H₂S 775 1142 9.417 Sulph. Dioxide 64.060 SO_2 17.030 8.856 731 1657 Ammonia NH 3 39.944 4.913 272 705 А Argon $C_p/Mol. = 9.420$ Tc 382 Pc 668 1.000 Mol. Wt. 19.53 TOTAL Minus 1.986 Cv/Mol. = 7.434

MOL. #

FRACT.

Cp/MOL.

EXHIBIT B

TC

PART

ΤC

PART

PC

571

58

PC

673

709

Cp/MOL.

FRACT.

 $\frac{Cp/Mol.}{Cv/Mol.} = \frac{9.420}{7.434} = 1.267 \text{ (n value)}$

S.G. =
$$\frac{19.53}{28.966}$$
 = 0.674

.





HEAT-CAPACITY RATIO (K Value)

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EXHIBIT E

COMPRESSOR HORSEPOWER SELECTION CHART

DISCHARGE PRESSURE (PSIG) (BRAKE HORSEPOWER PER MILLION CU. FT.)

														·····															
	25	50	75	100	125	150	175	200	250	300	350	400	450	500	550	600	650	700	750	800	850	900	950	1000	1050	1100	1150	1200	1 1 1
0	65	99	123	144	156	168	178	187	203	218	200																		4G
10	35	63	85	104	121	131	140	149	163	175		100	205	Ø.	2.3														L.
20		43	62	78	92	106	118	128	139	151	180	170	178		103	100		212	218	225									ļų
30		29	47	62	74	85	96	107	123	133	143	152	159	167	173	179	185	191	196		206	211	216	221	225	2.0			₫
40			36	50	61	72	81	90	107	121	130	138		152	158	Œ	170	175	180	185	190	1050 1	153	202	206	210	213	218	- 4
50			26	41	52	61	70	78	93	106	119	127	134	141	147	153	155	183	168	173	177	181	185	189	193	196	200	203	
60				32	44	53	61	69	83	95	108	118	125	131	137	143	143	153	158	162	166	170	174	178	182	185	188	192	ı
70				25	37	46	54	61	74	86	97	109	117	123	129	135	140	145	149	153	157	161	165	169	172	176	179	182	
80					30	40	47	54	67	78	89	98	109	117	122	127	132	137	142	148	150	153	157	161	164	167	171	174	ц. С
90					24	34	42	49	61	72	81	91	100	109	116	121	128	131	135	139	143	147	150	154	157	160	163	166	TA
100						28	37	44	55	66	75	84	92	100	109	116	120	125	129	133	137	141	144	148	151	154	157	160	i.
125							25	32	44	54	63	71	78	85	92	99	106	113	117	121	124	128	131	134	137	140	143	146	JAN N
150								22	35	45	53	60	67	74	80	86	92	98	103	110	114	118	121	124	127	130	130	135	, F
175									27	37	45	52	57	60	71	76	82	87	92	97	102	107	112	115	118	121	123	126	
200										30	38	45	52	58	63	68	73	78	83	88	92	96	101	105	110	113	116	118	
250											26	33	40	46	51	56	60	65	69	73	77	81	85	88	92	95	99	102	
300					ļ				•			23	30	36	41	46	50	54	58	62	66	69	73	76	79	83	86	89	
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400					ļ		L								25	30	35	39	43	46	50	53	56	59	60	64	67	70	E E
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700									ļ										ļ	ļ	ļ	22	25	28	30	33	36	38	
750									1														20	24	27	29	32	34	

EXHIBIT E

SUCTION PRESSURE (PSIG)



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EXHIBIT G



Dimensions and specifications

Sizo	Piston	Max,		Approx.	Gas	Appprox	cimate overall di	mensions
inches	cfm	psi	Rpm	ibs.	pipe, inches	Α	B	С
21/2 x 5	16.3	1500	675	1275	1 1/2	6'-5¾ "	4'-11⁄4 "	1'-10"
31/8 x 5	27.0	1350	675	1300	1 1/2	6'-5¾ "	4'-1 1/4 "	1'-10"
31/2 x 5	34.7	1350	675	1340	1 1/2	6'-6]16"	4'-1 1/4 "	1'-10"
4 x 5	46.2	1350	675	1390	2	6'-71/8 "	4'-21/4 "	1'-10"
5 x 5	73.8	1200	675	2135	3	6'-81/2 "	4'-15/8"	1'-10"
6 x 5	108.0	300	675	1815	3	6'-5¾"	4'-0% "	1'-10"
31⁄8 х 7	27.6	1500	514	1610	1 1/2	7'-3¾ "	4'-81/4 "	2'-7 1/2 "
4 x 7	48.0	1500	514	1700	2	7'-5½ ″	4'-91⁄4 "	2'-7 1/2 "
5 x 7	77.4	1200	514	2445	3	7'-61/2"	4'-85/8"	2'-71/2"
51⁄₂x7	94.6	1250	514	2670	3	8'-1! ₁₆ "	5'-1 ¾ "	2'-71/2 "
6¼ x 7	123.0	1250	514	2670	3	8'-11/16"	5'-1 3/8 "	2'-71/2"
8 x 7	205.0	300	514	2220	3	7'-13⁄8″	4'-4 7⁄8 "	2'-7 1⁄2 "
31/8 x 9	30.3	1500	450	2610	1 1/2	7'-9%	4'-10'14"	3'-3"
4 x 9	53.3	1500	450	2700	2	7'-1015,"	4'-11	3'-3"
4¾ x 9	77.4	1000	450	3250	3	8'-01/8 "	4'-101/2 "	3'-3"
5½ x 9	106.0	1250	450	3670	3	8'-4° ₁₆ ''	5'-0% "	3'-3"
6¼ x 9	138.0	1250	450	3670	3	8'-4316"	5'-0% "	3'-3"
7¼ x 9	188.0	800	450	4320	4	8'-4316"	4'-11%"	3'-3"
8 x 9	230.0	300	450	3220	3	7'-71/4 "	4'-6¾ "	3'-3"
9 x 9	293.0	550	450	4140	5	8'-11/2 "	4'-101/4 "	3'-3"
10 x 9	363.0	150	450	3750	5	7′-115⁄8″	4'-8% "	3'-3"
4½ x 11	73.0	1500	400	5180	21/2	10'-03⁄4 "	6'-27/8 "	3'-7"
5¼ x 11	102.0	1400	400	5000	3	9'-6"	5'-111/2"	3'-7"
6 x 11	136.0	1250	400	5230	3	9'-93⁄4 "	6'-0 ⁷ / ₈ "	3'-7"
6½ x 11	161.0	800	400	5830	4	9'-8°is"	5'-103/4 "	3'-7"
7¼ x 11	202.0	800	400	5830	4	9'-8'16"	5'-10¾ "	3'-7"
8 x 11	248.0	800	400	5830	4	9'-8%6"	5'-10¾ "	3'-7"
9 x 11	316.0	550	400	5650	5	9'-5"	5'-9¾ ″	3'-7"
9½ x 11	353.0	550	400	5650	5	9'-5"	5'-9¾ "	3'-7"
10½ x 11	433.0	550	400	5650	5	9'-5"	5'-93/8"	3'-7"
12 x 11	568.0	350	400	6230	6	9'-8"	5'-11"	3'-7"
14 x 11	776.0	150	400	6580	8	9'-10 ⁷ 16"	6'-0¾"	3'-7"
5 x 13	84.5	1500	327	6500	21/2	10'-7 1/2 "	6'-97/8"	4'-5½ "
6 x 13	127.0	1250	327	6550	3	10'-41⁄2 "	6'-7%"	4'-51/8"
7 x 13	177.0	900	327	7040	4	10'-4 % "	6'-4 1/8 "	4'-51/8 "
8¼ x 13	251.0	550	327	7160	5	10'-3½ "	6'-5¾ "	4'-51/8 "
9¼ x 13	319.0	550	327	7160	5	10'-31/8 "	6'-5¾"	4'-5½"
11 x 13	455.0	350	327	7500	6	10'-3"	6'-6"	4'-51/8 "
12 x 13	544.0	350	327	7500	6	10'-3"	6'-6"	4'-51/8 "
14 x 13	745.0	150	327	7950	8	10'-65/8 "	6'-7¾"	4'-5½ ″
17 x 13	1105.0	100	327	8100	8	10'-43/9"	6'-61/2 "	4'-51/0"

COMPRESSED AIR DATA

Areas of Circles

Diameters in Inches and Areas in Square Inches

11 012272 14 012272 15 110447 1935 1935 14 001322 1 7854 14 90402 14 12272 14 12272 15 14849 16 12672 17 7854	SXXXXXX XXXXX	44.1787 45.6636 47.1731 48.7071 50.2656 51.8487 53.4563 55.0884 56.7451 58.4264 60.1322	15 X8 X4 3/8 X2 5/8 3/4 1/8	173.782 176.715 179.673 182.655 185.661 188.692 191.748 194.828 194.828	1 X X X X X X X	471.436 481.107 490.875 500.742 510.706 520.769 530.95	1/2 1/2 1/4 1/2 1/2 1/2	1209.95 1225.42 1240.98 1256.64 1272.39 1288.25
1 012272 3 040087 3 110447 3 19635 5 441787 3 601322 1 7854 1 2849 4 1272 3 12272 3 13671 3 17671	8 121212 12121 121	45.6636 47.1731 48.7071 50.2656 51.8487 53.4563 55.0884 56.7451 58.4264 60.1322	15 15 14 14 14 14 14 14 14 14 16	175.785 176.715 179.675 182.655 185.661 188.692 191.748 194.828 197.939	25 25 25 24 25 26	481.107 490.875 500.742 510.706 520.769 580.95	74 12 14 40 14 1/2	1200.05 1225.42 1240.98 1256.64 1272.39 1288.25
24		47 1731 48.7071 50.2656 51 8487 53 4563 55 0884 56.7451 58.4264 60.1322	16	179.673 182.655 185.661 188.692 191.748 194.828	25 4 26 26	490.875 500.742 510.706 520.769 530.95	40 /4 /4 /2 /	1240.98 1256.64 1272.39 1288.25
10635 1065 1065	8 XXXXXXXX 8X	48.7071 50.2656 51.8487 53.4563 55.0884 56.7451 58.4264 60.1322	16 16	182 655 185.661 188.692 191.748 194 828 197 955	26 26	500,742 510,706 520,769 530,95	40 /4 /2	1256.64 1272.59 1288.25
1 306766 34 441787 4 6013927 1 7854 99402 34 1 2972 34 1 4849 34 1 4849 34 1 7671	8	50.2656 51.8487 53.4563 55.0884 56.7451 58.4264 60.1322	16 16	185.661 188.692 191.748 194.828 197.939	26 14	510.706 520.769 530.95	14 14 14	1272.59
34 441787 76 601382 1 .7854 99409 14 12272 14849 134 17671 14 .9700	, XXXXXXX	51.8487 53.4563 55.0884 56.7451 58.4264 60.1322	1/2 5/8 3/4 1/8	188.692 191.748 194 828 197 999	26 ³ ⁄4	520,769 530.95		1288.25
% .601592 1 .7854 ½ 99402 ½ 1.2272 ¾ 1.4849 ½ 1.7671 ½ 1.7671	, XXXXXX	53.4563 55.0884 56.7451 58.4264 60.1329	4 3/4 1/8	191.748 194 828 197 999	26	530.95	1 44	
1 7854 16 9940e 14 1.2272 36 1.4849 17671 17671	NXXXXX	55 0884 56.7451 58.4264 60.1322	16	193 828		F	1	1309.20
99402 1 1.2272 1 1.4849 1 1.7671	2X.X.X.X 9	58.4264 60.1322	16 1	14/ 933	74	541.19	41	1320.35
74 1.2272 56 1.4849 52 1.7671	9 9	60.1322		901 089		569 009		1000.90
1 7671	9 ³ 4	A1 0404	~ v4	204 218	27 74	579 557		1369 00
12 0 0000	9	01.0020	ý,	207 396	<u> </u>	583 201	42	1385.45
911 X.0739 I		63.6174	1	%10.598	14	593 959	14	1401.99
2 4055	- 74	65.3968	1/2	215.825	X	604.807	14	1418.65
1/6 2.7612	X	67.2008	1	217.077	28	615.754	1. 14	1435.37
2 5.1416	2	69.0295		220.354	신	626,798	43	1452.2
3.5466	21	79 7800	17 1	223.035		037.941 RAQ 199		1486 17
71 3.9/0L		74 6691	" v.	250 551	2974	660 591		1505 5
4 4 9087	2	76.5888	í,	233.706	- 'Y	671.959	44	1520.55
5.4119	10	78.54	1	237 105	1/2	683 494		1537.86
5.9596	1/1	80.5158	1/2	240.529	. 14	695.128		1565.29
76 0.4918	- X	82.5161		243.977	30	706.86		1572.81
3 7.0686	2	84.5409		247.40	2	780 419	4 5	1090.43
21 7.0099 1/ 8.9059	2	88 6643	18 18	250.848	3/1	742 645		1695 97
8 8 9469		90.7628	Γ.	258.016	31	754 769		1643.89
14 9.8211	12	92.8858	12	261.587	14	766 992	46	1661.91
10.3206	11	95.0334	1	265.185	14	779.313		1680 02
11.0447		97.2055	1.2	268.805	14	791.759	12	1698.23
11.7953		99.4022		272.448	32	804.25 918 985	17 24	1710.04
14 10 9841	12	101.0239		979 811	12	829 579	1	1758 45
14 14 1863	6/	106.1394	19	285.529	34	842 391		1772.06
15 035	12	108.4343	1/1	287 272	33	855 301		1790.76
32 15.9043	14	110.7557	1/4	291.04	14	868.309	48	1809 56
16.8002	12	113.098		294.852	3	881 415		1828 46
17.7206	21	117 950	2	298.048	14 14	007 029		1847 40
5 19 695	14	120 277		S06.355	~ 1/1	921 323	49	1885.75
14 20 629	13 I	122 719	1	310.245	14	934 822	14	1905.04
14 21.6476	1	125.185	20	\$14.16	_ 14	948 42	1/2	1924 45
22.6907	14	127.677		522.063	35	969 115	1. 34	1945 91
23 7583	. 18	130.192	13	330.064		975.909	120	1963.5
78 24.8595	13 14	198 907	51 ⁷⁴	000 104 44 941	3	1008 79	12	9009 97
12 27 1086		137.887	1	\$54.657	36	1017 878		2022.85
6 28.2714	1	140.501	¥2	305.051	- X	1032 085	[51 🗋	2042.85
1 29.4648	1/2	143.159	1. 14	371.545		1048 349		2062.9
30 6797		145.802	22	580.154	1, 14	1060.792	3	2083.08
71 51.9191		148.49		383.844	3″ 1⁄	10/3 113	82 14	T103.30
77 53.1831	4	151.201		406 494	1/2	1104 469	1 4	2144.19
35.7848	<u> </u>	156 7	23	415.477	14	1119 244	14	2164 76
74 37 1994	i 🥻	159.485	X ا	424 558	38	1154.118	34	2185 42
7 38 4846	14	162.296		488 787	14	1149 089	53	2206 19
16 59 8715		165.15	~ *	445.015		1164 159	1 3	2227.05
<u>141 9926</u>		167.99	4	45%.39	120 74	11/9.527	13	1148.UI
78 41 7184	74	1/0.0/11	1 74	401.004	"	1124.383		1100.01

EXHIBIT H

ELEVATION.	FT	2700
ATMOSPHERIC PRESSURE	PSIA	13. 296
SPECIFIC GRAVITY (AIR=	1)	0.674
N-VALUE (CP/CV)		1. 244
*****COMPRES	SOR DAT	ГАжжжжж
MAKE. WORTHI	NGTON	
MODEL HBGG13		
STAGE NO.		1
NUMBER OF CYLINDERS.		1
CYLINDER DIAMETER.	IN	6, 000
STROKE.	IN	13.000
RPM. A.L. States of the second second		327
ROD DIAMETER	IN	2, 500
DISPLACEMENT.	. CFM	127.038
CONDITIONS	OF SEF	RVICE
SUCTION PRESSURE	PSIG	240,000
SUCTION TEMPERATURE.	F	80
DISCHARGE PRESSURE	PSIG	1050,000
***************************************	RMANCE	+::+::+::+::+::+::+:
SUCTION PRESSURE.	PSIA	253, 296
DISCHARGE PRESSURE	PSIA	1063.296
DISCHARGE TEMPERATURE.	F	255, 512
COMPRESSION RATIO		4, 1978
AVERAGE CLEARANCE	2	8, 950
VOLUMETRIC EFFICIENCY.		0.7125
CAPACITY.	MOFD	2268
BRAKE HORSEPOWER PER S	TAGE.	196
TOTAL BRAKE HORSEPOWER	1 	196
ROD LOAD - COMPRESSION	I-LBS.	24145
ROD LOAD - TENSION-LES	n Na ann an an an	17682

EXHIBIT I

396